

# Airworthiness Directive Schedule

## Gliders

### PW-6U

26 July 2018

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- Notes:**
1. This AD schedule is applicable to PW-6U gliders manufactured under EASA Type Certificate (TC) Number A.088 held by Zakład Szybowcowy „Jeżów” (Previously manufactured under Polish TC Number GILC BG-213).
  2. The CAA has reviewed the WSK glider AD Schedule and split this schedule into three smaller AD Schedules. There is now an KR-03A Puchatek AD Schedule, a PW-5 Smyk AD Schedule and a PW-6U AD Schedule.
  3. The European Aviation Safety Agency (EASA) is the National Airworthiness Authority (NAA) responsible for the issue of State of Design Airworthiness Directives (ADs) for these helicopters. State of Design ADs can be obtained directly from the EASA web site at <http://ad.easa.europa.eu/>
  4. The date above indicates the amendment date of this schedule.
  5. New or amended ADs are shown with an asterisk \*

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**DCA/WSK/8 Control Rod Ends - Inspection**

- Applicability:** Model WSK PZL PW-6U gliders, S/N 78.02.07 through to 78.02.10 and 78.03.01 through to 78.03.03.
- Requirement:** To prevent failure of the aileron and airbrake quick disconnect rod ends, inspect per PZL-Swidnik Mandatory Bulletin No. BO-78-03-06.  
(AD SP-0085-2003-A refers)
- Compliance:** Within 5 hours TIS.
- Effective Date:** 25 March 2004

**DCA/WSK/11 Horizontal Stabiliser Rear Attach Fitting – Inspection and Repair**

- Applicability:** Model PW-6U gliders, S/N 78.04.01 onwards manufactured by Zakład Szybowcowy “JeŚów” Henryk Mynarski (ZSJ), and  
Model PW-6U gliders, S/N 78.03.08 through to 78.03.10 manufactured by Wytwórnia Sprzętu Komunikacyjnego “PZL - Świdnik” S.A. (WSK), and  
Model PW-6U gliders, S/N 78.00.00 through to 78.03.07 manufactured by WSK and fitted with an automatic elevator control connection embodied per Service Bulletin BS-78-02-04.
- Requirement:** To prevent failure of the horizontal stabiliser rear attachment fitting due to possible cracks in the attach fitting lug which could result in loss of the horizontal stabiliser and loss of aircraft control, accomplish the following:  
  
Inspect the rear attachment fitting of the horizontal stabiliser per the instructions in Zakład Szybowcowy “JeŚów” Henryk Mynarski Mandatory Bulletin (MB) BO-78-10-10 original issue, dated 07 June 2010 or later EASA approved revisions.  
  
If any cracks and/or damage is found, accomplish a manufacturer approved repair before further flight.
- Note:** The aircraft manufacturer has issued revision No. 03 applicable to the Aircraft Maintenance Manual which introduces the inspection requirements specified in this AD.  
(EASA AD 2010-0108-E refers)
- Compliance:** Before further flight, and thereafter after every hard landing and at intervals not to exceed 50 hours TIS.
- Effective Date:** 11 June 2010